

# AVIATION

AND

## AIRCRAFT JOURNAL

APRIL 11, 1921

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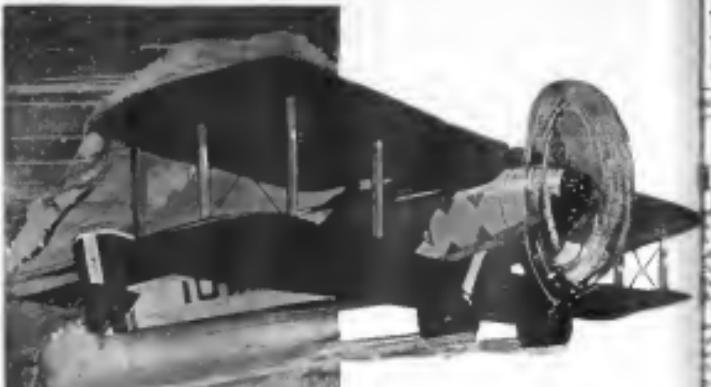
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# AVIATION AND AIRCRAFT JOURNAL

APRIL 11, 1921

No. 18



**W R I**  
AERONAUTICAL CORPORATION  
PARIS, N. J.  
Wright Aeronautical Division  
Model B. 1920. 7

**G H T**  
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POWER FOR ALL AIRCRAFT

## The Caproni Epsilon

THE Caproni flying boat Epsilon, which is shown on the front cover of this issue in many ways the most original type of aircraft that has been produced to date, is destined to be much in the great line as in general layout. In fact, as the matter of size and because the Epsilon, with a wing span of 300 ft. and an eight Liberty-22 engine, will be largely exceeded by the G.B.1, the first four boats ever under construction for the United States Army, for this will have a span of about 120 ft. and a power plant composed of nine Liberty-22 engines.

However, the general layout of the Epsilon differs fundamentally from the G.B.1—and from any other boat-type aircraft as well. To begin with, the hull of the Italian flying boat runs the entire length of the machine, a matter of 68 ft. On the hull, three cells are mounted, as a double tandem arrangement, these triplane cells with a total lifting area of 1,250 sq. ft. The forward and after cells are apparently mounted at the same height above the hull and are fitted with ailerons, while the middle cell is mounted on a lower level and has no ailerons. Several radiators are fitted in the after cell, but there is a tail, properly speaking, the rear ailerons being apparently mounted with fore-and-aft longitudinal control.

This arrangement, of course, is a radical departure from modern practice. The biplane flying boat, which originated with Prof. A. P. Langley's steam-driven airplane model, in 1889, and which that scientist later incorporated in his marine flying machine, has yet to demonstrate its efficiency. Its stability is readily conceded, but this does not necessarily imply controllability. Aerodynamic interference between following wings greatly detracts from their practical value and with the great and large parasite resistance to be found on the Epsilon this will represent an important item. Whether the bowing of the middle cell and with respect to the forward and aft cells will overcome this inherent defect of following-wing motion is to be seen.

The layout of the wing structure is of conventional design, with a series of wires interconnecting the cells. Along the middle wings of the cells two fairings run the entire length of the machine and house the power plant, which consists of eight Liberty engines. Four are fitted forward and four aft, with three frisees and three pusher propellers.

As far as streamlining and aerodynamic efficiency are concerned, this great boat, with its large number of wings and struts, and its numerous engine nacelles, is a retrogressive boat, perhaps it is too much to expect refinements of design as well as wholly experimental types. Apparently the designer's underlying thought was that by the use of these triplane cell wings and weight of lifting surfaces would be cut down to a minimum. At the same time, the weight in this type of construction being distributed longitudinally, this would further reduce bending moments and weights. The resulting decrease in structural weight explains the enormous

speed with which this machine is credited—100 passengers or fuel of 5,000 miles—and so may the adoption of small factors of safety. This seems to be borne out by the structural failure which occurred on the trials of the Epsilon.

Despite this initial failure, the further progress of the Epsilon type deserves to be watched with continued interest. However great the difficulties that must be overcome in order to make such a machine aerodynamic, nothing would be more stimulating than to witness the success of the experiment, for it would open up untried roads in airplane construction.

## Factors Affecting Maximum Speed

AS a rule it is customary to consider maximum speed as affected by the loading per square foot and the maximum lift coefficient of the wing. Quite a number of other points, however, enter into the determination of the maximum speed, and the recent N.A.C.A. report by F. A. Norton is highly interesting from this point of view.

The reported measurements that consideration be given to the effect of slip-stream, but also states that this effect is very small. The vertical component of the airframe thrust may also improve the landing speed. Reasonable air speed gives the pilot more confidence in coming right down to landing speed. Take above all, angle longitudinal and lateral control at low speeds are important, as when a plane is flying at a large angle of incidence, the ailerons become ineffective. The rudder is a little better off and so is the elevator. The more powerful all these controls are, the less danger is there for a machine going out of control at low speeds.

These various factors are very well brought out in the report and are well worth studying by designers.

## French State Assistance to Aviation

THE progress of the new French airship system for semi-rigid aircraft, which is planned in this area, are characteristic of the comprehensive manner in which commercial aviation is being fostered by the French government. Whether the merit or defect of state subsidies, it cannot be denied that France is making a bold and determined move with a view to basic expansion in the air. That it is proposed to achieve this result by means of commercial air supremacy instead of by the maintenance of a large air force, which would be impracticable in times of peace, surely shows breadth of vision. A semiprivate air transport fleet not only keeps pilots in training and thus affords in time of war a valuable reservoir of trained personnel; it also enables the government to assist a healthy development. As a consequence, on the beginning of hostilities the regular air force has the physical means for rapid expansion, which a country devoid of a commercial air fleet would naturally lack. The assistance the French government grants in commercial aviation is therefore a striking illustration of foresightedness.

# The Development of Aircraft

With Special Reference to the Zeppelin Airships

By P. Janay

Translated by Capt. Truscott, Aeronautical Engineer, Bureau of Construction and Repair, Navy Department  
(Continued from our last issue)

## III. Standard Representation of the Development

Comparisons will now be made, on the basis of figures obtained from practice, which will characterize numerically the development of aircraft in the directions indicated. The writer, however, has done this in a rather stephenerly manner, not for the present, but for the future, more completely. This appears the more necessary since the writer has, unfortunately, the Zeppelin airships—seems at present to be better adapted for transportation over intermediate and longer distances than the airships.

It is, however, possible that the margin of superiority will shift somewhat in favor of the airships with further development, but apparently the opposite is more probable. The following figures show plainly enough something which the author made known in influential quarters in 1914, namely that the economy of the airships from a technical and commercial point of view either decreases with increase of size or remains very nearly unchanged; while with the airship, the economy increases rapidly with increase of size. In the measured

analysis of the development besides the ratios  $\frac{1}{2}$  and  $\frac{3}{4}$

$\frac{1}{3}$ , and  $\frac{5}{6}$  as to general dimensions, power and speed are also required, consequently, these must be measured first. The following arrangement will be developed.

### DEVELOPMENT

#### (a) The Airship From a Technical point of view

#### (b) The Airplane

##### (1) Dimensions

Gas volume

Plane area

Power

(2) Speed

With 1, 2 and 3 the aerodynamic coefficient

(4) Useful load

With 1 and 4 the structural coefficient

(II) From a Commercial point of view

With the Frischke Method (Economy Coefficient)

II. The first beginning of aerial navigation—with wings of less than 10,000 sq. m. weight still less than 100 kg. per sq. m. The first beginning of aerial navigation is shown here. The beginning of the development of the airship can be connected with the first Zeppelin airship, which in 1900 had a volume of 20,000 cu. m. Hence the Zeppelin airships held the lead predominantly throughout the whole development. The data of the first airship will be briefly mentioned in connection with the first of the present sections.

The increase in size took place slowly at first. A volume of about 20,000 cu. m. was first reached with type A (LZ-10) and with the exception of type C (LZ-12) was kept until type M (the last ship of 22,500 cu. m., LZ-37). Then there followed rapidly one after another.

Type A (LZ-10) of 20,000 cubic meters at the end of 1914, type B (LZ-36) of 25,000 cubic meters at the beginning of 1915.

Type C (LZ-12) of 35,000 cubic meters at the end of 1915, type D (LZ-35) of 55,000 cubic meters at the middle of 1916, type E (LZ-36) of 65,000 cubic meters at the middle of 1916.

Even if these increases in size are to be summarized principally to the demand for greater altitude in the military ships, nevertheless there was thereby created aircraft of greater carrying capacity, which could, in a similar manner study the new possibilities of a means of transport, employing a large useful load with a large range of action. Type E moreover actually had to fulfill in war time the function of a cargo airship, and the journey of the L-39 (LZ-194) from Jassfeld, (Bohemia) to the former German East Africa

without taking on any fuel and in about 200 hours out of 240—very plainly showed its suitability for the purpose.

The other airship building concern can also reach a maximum in the volume of their ships but they reached these maximums to those just mentioned. The British, for example, in the beginning of 1919 brought out the type B-1 and B-2, which, as the measure of the economy of the German airship, was referred to in many papers as the largest type airship, although they were of only 50,000 cu. m. volume.

The largest Zeppelin airship that came before the Boisheim airship was the C-10.

It was built of 20,000 cu. m. gas volume for long-distance flights and had a range of 1,000 kilometers.

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be regarded only as an experimental ship. Type  $\pi$  reached 11.8 although the resistance coefficient was lower than 1.0. The already somewhat flattened form may have caused a decrease in the coefficient of resistance.

In type  $\delta$  the aerodynamic coefficient decreased somewhat—10.8 or 12.4 after the installation of a more powerful engine. In type  $\pi$ , which undoubtedly had smaller parasite resistance, it was again 12.3 although the form of this ship certainly was an improvement over type  $\delta$ —the coefficient of friction decreased markedly and from an aerodynamic point of view the last step is clear.

A comparison of the results with type  $\pi$  in this ship—the results in the last form discussed. The form was somewhat fatter, as with the same engine, it was considerably shorter. Also the rounded surfaces, which from this type were fitted only at the stern, were reduced so that the aerodynamic coefficient of 23.4—16 per cent of the increased engine power, which the propellers used were not suitable—was unchanged.

Type  $\pi$  is only slightly better than  $\delta$  in spite of the improvements in the parasite resistance, especially in the rounded surfaces. It is also somewhat more slender. Type  $\pi$  reached the value of 28.6 which, however, after the fitting of more powerful engines (from 12.5 to 13.5) decreased somewhat again to 26.6.

Types  $\pi$ ,  $\alpha$ , and  $\beta$  (including L2-32) have low aerodynamic coefficients throughout.

Only with type  $\pi$  does the coefficient, rise again to 27 which without doubt may be attributed principally to the improvements in the hull and rudder. (From 26.6 to 27.0) Type  $\pi$  is the first for the first time with a hull driven by a propeller at the first end of the forward part, instead of two propellers, showed no aerodynamic improvement. It had a value of only 25.9—because the work of the forward propeller swept over the midships cabin with which this ship was fitted, causing an increase in resistance. On account of the removal of the midships cabin in type  $\pi$  the aerodynamic coefficient increased to 27.7.

A very notable increase from an aerodynamic point of view was obtained with type  $\pi$  which had a form similar to that from the surface ships with a much smaller length/diameter ratio. This form, based upon aerodynamic studies by the author, is found in the first two ships of the "Dolgoruky" class. While for the first ship the area of smaller hull segments together with smaller coefficients of resistance because of the increased faired and relatively smaller surface. With a structurally similar arrangement of type  $\pi$ , compared to 30.7 in type  $\alpha$ , the aerodynamic coefficient of 28.6 in the first of more powerful engines and the same propellers (from L2-32 to 34) the aerodynamic coefficient naturally decreases to 26.6, to again increase in type  $\pi$  to 35.1 on account of the continued reduction of the parasite resistance, and in spite of lengthening.

Type  $\pi$  which in a further reduction of the longitudinal resistance took place shows again a reduction of the after propeller with the uncorrected resistance, this figure increases in type  $\pi$  to 32.0 and in type  $\pi$  to 46.6, in which for the first time the side brackets on the after part were also completely removed and in their place a two engine installation was fitted, with a single propeller. In this ship also the stern cover was fitted, a single propeller. The large amplitude of today later showed 2 tons useful load, the final displacement—4.5 to 5.0 tons.

A further improvement from the aerodynamic point of view was again attained with type  $\pi$ , with a coefficient of 54.2, in which there were fitted special propellers based upon extensive studies. These propellers were in every respect with the same engine power, speed and the resistance of the ship. The comparison of speeds given under 1.3 shows that

Type  $\pi$  gave a small increase of aerodynamic coefficient—increasing it to 56.0—which, however, after the fitting of more powerful engines with the same propellers, again had the same resistance power, speed and a coefficient of 54.6. Type  $\pi$  in which the same propellers were used as in the type  $\pi$  ships with the coefficient of 56.0—attained however only 54.4 on account of the increase in length of this type.

A very important step in the improvement of the aerodynamic properties was made possible in type  $\pi$ , in that by the use of four direct drive propellers the resistance of the ship

and their propellers could be reduced, so that with the slight increase in engine power and the consequent increase in speed an improvement of propeller efficiency was secured. The aerodynamic coefficient of 62.2 reached with this ship makes the improvements most apparent.

Type  $\pi$  (the passenger ship "Dolgoruky") and the third, length of the model ship had slightly exceeded the last ship with a coefficient of 62.0—on account of further reductions in the hull—slightly reduced and from an aerodynamic point of view the last step is clear.

The comparison of the figures 7.7 and 62.0 gives a general picture of the development of the ship from an aerodynamic point of view.

If only Zeppelin ships have been discussed this far, it is because the figures compiled from ships from other yards are mostly much smaller. For instance the value of  $\frac{1}{C_D}$  for the other German ships, which only 40 for non-rudder, while the best English ship 11-31 (and L-34) shows only 47 as an aerodynamic coefficient.

The value  $\frac{1}{C_D}$  which can be used for comparison with the analogous figure of the airplanes is about 2 for type  $\pi$ , about 10 for type  $\pi$  and 28 for type  $\pi$ .

On the aerodynamic coefficient has not increased with the development of airplanes at the rate it has in ships. The Wright biplane of 1908 already had about 12 as  $\frac{1}{C_D}$  while the airplanes of 1914 could show only from 18 to 24 for non-rudder and 28 to 32 for rudder. The best biplane of 1914, the Wright biplane of 1915, shows only 30—26 to 32 can still be taken as normal.

In comparison with this, however, the development is shown much more plainly by the value of  $\frac{1}{C_D}$ . The Wright biplane had a coefficient of hardly 1, while with modern planes a value of 5 to 8 can be obtained.

This figure also permits a comparison with the corresponding coefficient for the airplane which, as was mentioned above, already stands at 28.

1.4. The development of airplanes shows a very remarkable increase with their development, with type  $\pi$  the resistance increases to type  $\pi$ —5.4 tons, in type  $\pi$ —12.2 tons, in type  $\pi$ —28.6 tons, in type  $\pi$ —49.6 tons. Type  $\pi$ —the cargo ship—naturally maintained—but the greatest useful load than the biplane.

If the useful loads of the airplanes are now compared with the figures given above, model, even though these airplanes also have been an extraordinary case. The first Wright biplane in 1908 could provide only about 70 kilograms of useful load. After many improvements the same type could 250 kilograms in 1909. Faired in 1910 could 1,000 kilograms. Stepped in 1911, was about 3,012—1,000 kilograms and in 1916—1,400 kilograms. The large amplitude of today later showed 2 tons useful load, the final displacement—4.5 to 5.0 tons.

THE STRUCTURAL CONSEQUENCES

A correct idea of this is offered by the representation of the structural coefficients, the ratio of Drafted Load to Weight Factor. Type  $\delta$  shows the smallest structural coefficient, while after the reconstruction was 0.95. This increased in type  $\pi$  to 0.96, in type  $\pi$  to 0.94, in type  $\pi$  to 0.94, while types  $\pi$  and  $\pi$  had reached 0.93, with 0.92 and 0.90 for  $\pi$  and 0.89 for  $\pi$ . Type  $\pi$  reached 0.84, while type  $\pi$ , in spite of some changes in the hull, was 0.83. The hull of the same was 0.84 and 0.83 with hull type  $\pi$  with 0.85 and type  $\pi$  with 0.84 were somewhat poorer. Type  $\pi$  with a greater gas volume and a coefficient of 0.72 was again considerably lighter. The small ships of type  $\pi$ , however, as a result of many changes in the hull reached above 0.92. A further increase in the structural point of view brought type  $\pi$  with 0.70 while type  $\pi$ , which was a lengthened type  $\pi$ , reached only 0.78.

The most notable increase of the structural coefficient was reached by type  $\pi$  and its derivatives. These ships, of types  $\pi$  to  $\pi$ , had structural coefficients of 1.0 to 1.6 while type  $\pi$ , the largest ship of the series, reached 1.6. Type  $\pi$ , in spite of a somewhat lighter than  $\pi$ , had the same structural coefficient as the latter, 1.6, because it was designed for considerably greater engine power and consequently was built relatively heavier. Finally the small type  $\pi$  (about the same size as  $\pi$ ) which had a structural coefficient of 0.861 shows with its structural coefficient of 4.0 the improvement from a structural point of view which the airplane plane since 1912.

The comparison of 9.5 type  $\pi$ , and 3.6, type  $\pi$ , summarizes the structural improvement as a whole.

A comparison with the average of other ships is also of interest here. The best of these ships already reached a structural coefficient of 6—1.6 for rudders in Germany, 0.6 for rudders in England, 0.65 for non-rudders in France, and 0.9 for rudders in Italy.

(1) At the beginning of this development the coefficient  $\frac{1}{C_D}$  for airplanes had a value of 0.5. In 1914 the value increased still somewhat over 0.5. Today the passenger planes are of about 5 to 7. Exceptionally highly constrained craft, e.g. the British small airplane, attain 3.5 and more. (The 40 hp. Bleriot 1914 had to have a coefficient of 1.21 and was 10 tons, Wright 1916, 1.06.)

As far as can be learned from available information no type of very large or giant airplanes attains 6—0.85.

#### THE FIGURE OF MERIT

(a) The values of the "Figure of Merit"  $\lambda$ , obtained from the product of the aerodynamic coefficient and the structural coefficient will be given in enclosed form.

If we begin with type  $\pi$ , which after the reconstruction had a value of  $\lambda = 1.6$ , the most important types show the following numbers: type  $\pi$ —0.5,  $\pi$ —0.6,  $\pi$ —1.6,  $\pi$ —1.6,  $\pi$ —1.6. Thus follow the subsequent types  $\pi$  with 0.95 and  $\pi$  with 0.97, with 0.92,  $\pi$  with 0.94,  $\pi$  with 0.93, with 0.92 and 0.91.

This shows the value of the figure of merit from 1.6 to 0.93 (the whole development of the Zeppelin ship) from a technical point of view.

As neither the aerodynamic coefficients nor structural coefficients of the ships of other navies plane reached the values of these figures—on the contrary, a noticeable increase of the value of the resistance in the ships of other navies is to be expected of the resistance in the ships of other navies. These best stand at between 30 and 60 per cent of the best value of  $\lambda$  of the Zeppelin ships.

The fall of the value in 1908 in type  $\pi$  is to be attributed to the greater weight of the propelling machinery. Type  $\pi$  hardly shows a value of about 0.62 because it again increased, and therefore, and therefore, the value of the same was increased, however, larger compared.

If we write:  $\lambda = \frac{1}{C_D} \cdot \frac{1}{C_S}$   
then for type  $\pi$ :  $\lambda = 0.5$

(b) In comparison, airplanes at the beginning stood at about  $\lambda = 0.5$  and now are between 25 and 40  $\lambda$ , now, however, between 10 and 15  $\lambda$ , which is about 1/3 of the corresponding value of the airplane.

#### THE ECONOMIC COEFFICIENT

(II-1) As an indication of the development from an economic point of view the economic coefficient as was derived. It should suffice if the ship and airplane types of the lowest cost of construction and cost of the engine are compared, a maximum of time—use is selected for determining the value of this coefficient.

For the purpose of comparison a load wind of 50 km/hour will be assumed for  $W$  in  $m^2$  (299 and 360), in form the necessary of making a profit in such common load would not be taken into account.

The type  $\pi$  Zeppelin ship has a value of  $\lambda = 0.5$ , that is, it is about 10 per cent of exceeding its operating costs. Consequently it would be impossible to load profitably with this type  $\pi$  when built and operated today. Like the type  $\pi$  the  $\pi$  would also come off badly today even with the twelve best waggons full capacity which was assumed. It is twelve times  $\lambda = 0.95$ . Even the need to the best Zeppelin

ship, type  $\pi$ , the Sachsen, would have today, with  $\lambda = 1.67$ , as model probabilities, since this value would qualify rail traffic with short daily stops in not quite full capacity.

The  $\pi$  Zeppelin type  $\pi$  would be only better off with  $\lambda = 1.65$ , even though with a daily stops of the same size as the Sachsen, which, since the Delag at present has only one ship in service, to actually the case—falls to about 1.6. The largest type  $\pi$ , with  $\lambda = 0.93$  in model better, but the fact as far as the cargo ship, type  $\pi$ , with  $\lambda = 0.93$ .

II. No airplane service today can result similar figures as the same ratio.

From the figures which have been given it is evident that with increasing use of numbers of all the coefficient increases, and even in very rare instances—in which the data usually is improbable.

#### Conclusion as to the Further Development of Aircraft

From the figures which have been given it is evident that with increasing use of numbers of all the coefficient increases, and even in very rare instances—in which the data usually is improbable.

The following reasoning gives a further analysis. The  $\pi$  increases as the third power of the linear dimension. The



FIG. 2. ECONOMY COEFFICIENTS FOR AIRSHIPS AND AIRPLANES AT 150 KM/HR. (200 REFERRED TO WRIGHT 1916)

will weight increases as some power less than the third must they depend partly on the engine, partly on the surface and partly on the transverse section. The power plant weight for the same speed increases as the second power of the linear dimension, which depends on the resistance. Consequently these results are an inverse of the useful load at some power relatively greater than the ship.

Apparently there is no upper limit to the increase in size of the ship. To be sure, loadings will be more difficult with ships of larger size, but the problem doubtless lies within the range of a technical solution even for ships of very great size. The loading and operation of the ship is the same as for the airplane. The construction, on the other hand, will be more difficult with ships of larger size.

The dependence of resistance on the measure of size has been plainly shown by the reasoning above. The curve of Fig. 2 shows this very suitable increase of the economy in relation to the weight savings. It will be seen that the probabilities under the assumed conditions begin to be positive at a weight empty of about 8 tons (that is about 8000 m. m. in  $m^2$  of the wing surface). The economy of the airplane is 100 per cent and before 180,000 m. m. volume it reaches 500 per cent.

The circumstances are entirely different with the airplane. At first sight and as we see the figures which have been given it seems that an improvement is possible with increase in size. This impression corresponds to the facts. If a train of reasonable number to the one given above is followed there remains for the airplane an approximately equal increase in the size of the wing surface. Consequently it would be impossible to load profitably with this type  $\pi$  when built and operated today. Like the type  $\pi$  the  $\pi$  would also come off badly today even with the twelve best waggons full capacity which was assumed. It is twelve times  $\lambda = 0.95$ . Even the need to the best Zeppelin

increasing the size of the airplane. (See following II.)

and III). Considering the nearest possible equal factors of safety II should be taken as the exponent for passenger carrying aircraft.

Finally, landing gear and engine weights likewise increase at the rate somewhat higher than the square, but less than the square. The power plant weight for the same speed would increase slightly as the annual power if the unit weights of engines, radiators, piping and controls remained the same as the plane increased in size. Thus, however, according to experience the first, the increase in weight of the plane is far greater than the increase in the annual power. Thus, consequently requires for the useful load an increase of a rate considerably less than the annual power.

It would be admitted that plainly marked upper limit to the airplane is not fixed, although it can already be perceived that the increase in size must result in an improvement. But the landing differences increase with increase in the size of the airplane. Thus, says the author, it is to hardly believe that it will be necessary which would be the case in certain cases in the case of the airplane with its landing speed of above zero. The landing and operation of the plane in the air will probably be more difficult with increase in size, an increase of the greater forces involved in disturbance of stability, and in its case, noise.

In the degree of safety the airplane which starts a horse to the increase in size. With planes constructed as at present this has far below the most recent size. The curve of the economic coefficient as plotted above the weight capacity—as shown in Fig. 9—rises rapidly at first reaching 1, under the conditions assumed with a plane of very small size. It exceeds 1 only a little, reaching the maximum of 1.22 at about 12 tons weight capacity (assuming a landing speed of 20 mph) and then begins to decrease, reaching 0.95 at about 20 tons weight capacity (assuming a landing speed of 20 mph). The plane is not yet considered, that is, piled and five or six passengers, for a trip about 1200 km) and from these sinks again below 1.

If the two curves as and as are compared the economy of aerial navigation by the airplane, from the viewpoint of technical perfection, is beyond doubt with the exception of the important differences noted.

Thus, as regards the use of the proposed development of the airplane, although it is of course still capable of very material improvement. There is room for the airplane. The airplane designer is to be very lenient on his basis. It is possible for him to design and build airplanes of 20,000 lbp. These will tell if it is worth while, provided the business man does not interfere.

A great task still awaits the airplane designer. For, as it increases in size, the airplane will become more efficient than it has been attempted to show here, if it is further improved and perfected. The lines along which this should proceed are known.

On the basis of the general relations for the conditions of equilibrium coefficients are derived to characterize the development of aircraft from a technical point of view. Numerical values are obtained for these from a copious supply of data. In this there are distinguished in the separate investigation of stability and strength.

#### Popularity Efficiency

The Air Service Mechanic School at Kelly Field, Tex., was recently transferred to Chaffee Field, Ranchos, Calif., where work continues on Jan. 27, and was transferred on Feb. 1, 1941, to a portion of MacCoy Field, the new economic of Chaffee Field. It took nearly over a mile just a short time, to move the entire equipment of the school to its new quarters.

Flying has been carried on since Feb. 1 at this field only as the locality being very favorable for any kind of flying.

#### W. H. G. Aircraft Corp.

The W. H. G. Aircraft Corp., Inc. of 52 West St., New Haven, Conn., a new company, has been organized, recently, for the production of aircraft which will be used primarily as a former Government Aircraft Inspector, and by Lt. Col. M. Cummings, a former U. S. Army pilot, Secretary and Sales Manager.

The factory of the new company is conveniently located near a small flying field and also near the bay of Port of New Haven, which makes it a good sea-air port for local firms.

magitude of economy only the small airplane will remain a real competitor of the small ship. Against the small airplane no competitor can compete, least of all the giant airplanes.

#### Protecting the Public

The following is a reproduction of a warning notice issued by the British Air Ministry and which is displayed on and in the neighborhood of aerodromes in Great Britain:

## WARNING TO THE PUBLIC IN REGARD TO AIRCRAFT

### WHEN AN AEROPLANE IS ON THE GROUND

DO NOT stand near the machine—the pilot must use what he is doing.

DO NOT touch any part of the airplane, or you may endanger the pilot's life. Do what he tells you at once.

DO NOT stand at these lighted aerodromes within 20 yards of the machine; there may be dangerous pistol fire.

### WHEN AN AEROPLANE IS LANDING OR RISING

DO NOT run to where you think it will land. Keep out of the way near a hedge or other obstacle, and wait until it has landed.

DO NOT stand at the direct line of an aeroplane which is about to rise.

DO NOT to children or animals leap at the nose of a landing or rising aeroplane.

### IN CASE OF ACCIDENT

DO NOT be afraid, get out at once as quickly as possible.

Telephone or send for doctor and ambulance. Do not give any information.

Telephone or send to nearest telephone, giving number of machine and position of accident.

If the machine is burning, try and subdue flames with sand, earth, wet clothing, etc., or fire extinguisher if available.

Leave someone to guard the machine, if possible, and return home.

#### Stability and Strength

#### (3) An aerodynamic coefficient

Coeff. of Resistance  
Useful Load

#### (5) A structural coefficient

Wing Empty

#### (2) A figure of Merit

— the product of the other two.

To facilitate the development from an aeronautical viewpoint, figures have been obtained from the estimation of probabilities, which reduced to simple fractions are finally expressed as an aeronautic coefficient in absolute units.

From the computed values, which are reproduced in the form of curves, conclusions are drawn as to the further development of aircraft which show a considerable propensity for the growth of size, which however, is to be expected in the larger aircraft in which destroy the exaggerated hope, which so many persons have transported to the world, of the giant airplane as the method of transport of the future.

Even though the figures and limits assumed in this work change and shift somewhat in the course of time, and in the place of time or they something different can be substituted, in

# "Who's Who in American Aeronautics"

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#### Albert John Hall

EDWARD JOHN HALL, Aeroplane Mechanic, 1000 Main St., New Haven, Conn., April 19, 1941, married, 1937, Edward Hall, Mechanic, 1000 Main St., New Haven, Conn., March 1937.

Received City College Grammar School, New Haven, Conn., 1926, and City College of New Haven, Conn., 1930.

Worked City College Grammar School, New Haven, Conn., 1926, and City College of New Haven, Conn., 1930.

Worked in New Haven, Conn., 1930.



or fifty thousand feet in the air. The theory which Gladstone expounded, the old common law theory, was here applied analogically, that land and sea have either upwards or downwards, that it extended from the middle of the earth to the middle of the heavens.

#### A Modern Flap

The modern conception of air freedom and state sovereignty, as expressed at the report of the air committee on aeromaritime at the Peace Conference, is an absolute sovereignty of the air-space over land and territorial waters free from any right of innocent passage. This theory denies to foreign aircraft in territorial air-space even the right of passage vessels now obtain in territorial waters. A British vessel bound for Jacksonville from New York may not advance her hull within the Florida coast by sailing within the territorial waters of the United States, that is within the three-mile zone. This is exact. The British vessel has the right to use our territorial waters or any other waters, as long as it flies on peaceful commerce. International law gives her that right. The aircraft, however, presents a different problem. It involves a more intricate right, as the aircraft is not within the land. A vessel sailing the territorial waters of a state comes only within reach of the extreme edge of the state, whereas the aircraft flying overhead covers the interior of the underlying country.

The analogy of the air-space to the sea is not accurate. The sea is only adjacent to the land, whereas the air-space completely covers every foot of the land. Only the edges of a country are adjacent to the land, the further the land goes, the further he is from land. A country's frontiers fly miles out to sea to do little damage to a port. Neither can it obtain at that distance much information of any value. An aircraft, on the other hand, still remains over the land, no matter to what altitude it rises. The entire state has moreover to the air what the coast has to the sea, as much over a state as a coast a thousand feet up, and like the coast the sea is the air can still therefore the underlying country. But the fall from an altitude of five miles are quite sure to land somewhere. To be sure, commercial aircraft powerfully winging their way over a foreign land, will not necessarily threaten the safety of the underlying country, no more so than the tramp steamer threatening the safety of the country along whose shores it happens to be navigating. The air with the aircraft, however, has been designated as "territorial" by the different states, and for the present, it must go without. Before the World War partial freedom of the air seemed at least possible, if not now, not at the present time with the size of war still in the air, and with the consequent development of intense nationalism, all freedom theories have been swept aside for the controversial doctrine of full state sovereignty.

#### The Right of Innocent Passage

The doctrine of the right of innocent passage was nearly universally upheld before the war. It was upheld by the Institute of International Law and by the international legal conventions called to consider the subject of aeromaritime. A question in those days could rise from this banner, covering the hemisphere, could be easily raised over the matter. So long as the air was believed to be a natural and a suitable height, he was permitted to sail and to soar. The world's most maritime territories, such as Britain, but in the main the country was wide open to innocent passage. This was not an end to all evil flying. Only military aircraft were permitted in the belligerent states, and all flying was completely under surveillance. The fact that the different states have remained to a great degree the same, does not mean that innocent passage may again be permitted as in ages. International law rules have already been established, but that arrangement is more of an official nature between the individual governments, rather than the starting of innocent passage. Endured with her damages will not readily support the demand of freedom of the air-space. Trade, for full aeromaritime, is not other than, upheld the right of innocent passage. The International Flying Conference of 1919 provides for the innocent passage of aircraft. Article 2 states: "Each contracting state undertaken in time of peace

to assured freedom of innocent passage above its territory and territorial waters to the aircraft of other contracting states." So long as innocent passage is limited to "contracting states" there will be no universal freedom of innocent passage, but that is at least a start in the proper direction.

It is agreed that a state may prohibit flying over certain areas for the protection of its soil, or for military reasons. Such restrictions are natural and are not to be regarded as any further restrictions, however, will create a lot of bad feelings in the other states. The war created a number of states with no maritime boundaries, such as the present-day Austria, Hungary and Czechoslovakia. There are other states which have no sea routes, such as Switzerland, Bolivia and Paraguay. If air passage is only to be had over native territory, or over the sea, then the aircraft will be compelled to fly over land, except at the will of the underlying states. This involves a serious problem. One half step backstepped from carrying a party at Czechoslovakia from crossing German air-space by flying Parrot. Article 6 of the International Air Conference provides that no contracting state shall, except by a special international agreement, prohibit the crossing of the territory of an aircraft which does not possess the nationality of the contracting state. In other words, states not justified in the sovereign will be deprived of the common right of commercial international rights, one of which is the right to peaceful commerce, and to deprive the citizens of one nation the right to carry on their natural and usual occupations would be to deprive them a certain portion of common freedom.

The entry of all nations into the International Air Conference, in time, will undoubtedly help to solve the question of sovereignty and air freedom. The individual state must remain still and complete sovereignty of the air-space over its territory, but at the same time the air-space must be closed to through travel. In time of peace, of course, commercial aircraft, International Air Conference must be closed to through travel. International presence of aeromaritime laws, like the International Air Conference, as well as the International Aeromaritime Committee on Aviation in 1926, have declared themselves in favor of regulated international air regulation, with absolute state sovereignty as a national policy. The next step in the progress of aerial law will come about when the Conference recognizes the principle of innocent passage as an international policy.

#### Experimental Research on Air Propellers, IV

R. A. G. A. Report 185

This report is a confirmation of a report on the same subject published in the Fifth Annual Report of the N.A.C.A. The research was conducted in the aerodynamics laboratory of Lehigh University, and the report was prepared under the direction of Dr. W. F. Durand and Prof. E. P. Lesley. The report states the results of the investigations upon various propeller models, and the report of the N.A.C.A. on Aeromaritime and commercial aircraft. It is those interested in the design of air propellers. The discussions accompanying the report or necessarily somewhat brief, as the report is to be a part of the general report which will include a review of all the propeller investigations that have been conducted at Lehigh University. This general report will be ready for publication with the Seventh Annual Report of the Committee.

A second Report, No. 186, may be obtained upon request from the National Advisory Committee for Aerodynamics, Washington, D. C.

#### Air Service Flies Large Order

The U. S. Army Air Service on April 1 placed orders for 206 Thomas-Morse MB-3 pursuit airplanes and spares with the Boeing Aeroplane Co., of Seattle, Wash., and for 202 Martin bombers and spares with the L. M. B. Engineering Co., of College Point, Long Island, N. Y. The value of these orders aggregated nearly \$2,500,000.

## Balloon Racing\*

### A Game of Practical Meteorology

By Ralph H. Upson

Transportation has always been more or less dependent on weather. During the last fifty years, this dependence has been increased immensely by the number of aerial and space vehicles. Now, however, the advent of commercial air navigation makes weather more of a factor than it has ever been before, so much so indeed that the success of aircraft for commercial use will almost depend on how far the design program can be paralleled by developments in three-dimensional meteorology. It is the purpose of this discussion to study this very important factor of weather as it applies to aerial navigation. Meteorology is no more an exact science than medicine is. We are sure, there are laws and principles that can be logically related upon, but the great bulk of our future development for some time to come must depend on the accumulation and correlation of such facts, experience, and good luck. The importance of any aircraft, either commercial or military, is in the knowledge of its strength. (a) The power plant of the craft, and (b) the surrounding air, or, broadly speaking—the weather. Quantitatively, the effect of the weather is usually unchanged by a difference of speed and mass velocity of the aircraft. For example, a 20-mile-per-hour wind blows a propeller at varying strengths at just 30 m.p.h. regardless of the power of the engine.

As in other branches of science, the best way to study this important subject from a practical standpoint is to separate it so far as possible from academic interests which only distract the observations and confuse the results. The free balloon is almost ideally suited to one present purpose for the following reasons:

1. Being so light, its control is entirely dependent on aerodynamics with existing weather conditions. The performance of a balloon is exactly like that of a free particle of air with the addition of altitude control.

2. The entire freedom from pitching, vibration, noise, and wind, permits the most delicate observations to be made.

3. It is simple and safe to operate a balloon in a balloon basket, a device which is a sort of inverted canopy of a hot-air balloon. It is safe that it is practically fool proof. It would take considerably more than an ordinary fool to hurt himself in one.

The progress in other branches of aerometeorology has of necessity increased the value of the free balloon for testing. During the last ten years, many flights and a large proportion of aerometeorological observations have been made by free balloons. For training in navigation and meteorology, it has also been advocated for airplane pilots. But its greatest and broadest value lies in the general stimulus to meteorological knowledge to be gained by its development as a recognized sport. And it is no disparagement of the airplane that the free balloon is the sport that can be mastered. It is also very moderate in cost. Recent developments in fabric and gas generators put ballooning within reach, financially and otherwise, of any moderate-sized club.

The only piloting of a free balloon is easy and quickly learned. There are only two controls, balloon and gas. To go up, one pulls on the gas, to go down, one lets go. To stop going up, one pulls out. The control of altitude and width of the basket, the desired wind currents depends only on the proper expenditure of balloon and gas.

Ordinary flights usually last from one to twelve hours, the landing being planned for a time and place that will best suit the convenience of the passengers. The prearranged landing place is usually a flat, open, level field, which must constantly match his winds and still against the prevailing wind conditions, which are never twice the same. An interesting note, as well as a scientific treatise could be written about every balloon flight that was ever made.

The highest cut of ballooning finds expression in the national and international races for distance which are held

every year. These commonly run anywhere from 400 to 1,300 miles distance in the last year. I would not be at all surprised to say that one of these races will draw out almost every nation that has basic knowledge of aerometeorology, experience in its application, ability to read up the actual conditions, good judgment in their interpretation, persistent skill in handling the balloon, dreams in adhering to a good plan of action but always with eyes and wind open for a



U. S. NAVAL AIR BALLOON ENTERED IN THE 1928 GORDON BENNETT RACE

Sojourner, enough to sustain where necessary, and plenty of gas and ballast. The all round racing pilot, including great speed, skill, and ballast, observes, observes, observes, and makes the most of the opportunities he has. The control of altitude and width of the basket, the desired wind currents depends only on the proper expenditure of balloon and gas.

Our most immediate concern now is to see that America is well represented in this year's race. We are advised a total of three teams consisting of pilot and aid for the three different classes, each sailing being limited to the same number. The only way to organize is to cooperate for a purpose, the only way to compete is to compete for a purpose, and the only way to win is to win for a purpose. The difference concerns parts of the plan must include attention to the following:

1. Fostering the application is outside the scope of the



power or motion contained in said apparatus. The word "flight" shall include every kind of locomotion by which "A house, established, recognized field or place of landing" shall mean a public or private field or place of landing for such purpose, and "an emergency place of landing" is designated by the Bureau of Navigation as "an emergency place of landing" in any place where a landing may be effected in an emergency without endangering in any way life or property on such place of landing. "Landing" is in connection with the Fleet, but has given a further demonstration of the practicability of large aircraft for commercial traffic.

**Section 3.** No person shall go into any demonstration of flying, flying or aerial acrobatics, or give any demonstration of the methods which may tend to divert the aircraft from a normal flight with every consideration for safety and safety.

**Section 4.** No person shall fly over any part or section within the limits of the City of Chicago at a height lower than 1,000 feet, except in a place where it may be necessary and at all times in a leisurely, deliberate, or reasonable way, or in a deliberate place on land or water, and shall not perform under any circumstances, special or otherwise, fly at a height lower than 500 feet except at the beginning or end of a flight.

**Section 5.** No person shall attempt to pass or permit to be passed, or carried, any aircraft, or any aircraft parts, tools, instruments, handtools, cameras, used or other similar whatever, unless it is done over a place established for that purpose, and if equipment carried in aircraft shall be adequately fastened in place before leaving the ground.

**Section 6.** Aircraft approaching each other land on, or near, land, or water, in a safe, orderly and deliberate manner, their course to the right to pass and under all other conditions the aircraft bearing the other on its right shall keep.

**Section 7.** Lighter-than-air craft shall at all times have the right of way over heavier-than-air craft.

**Section 8.** Every person receiving the possession of these regulations shall be liable for damages, and upon conviction thereof shall be punished by a fine of not less than \$500.00, nor more than \$500.00, or imprisoned on the workhouse for a term not exceeding one year, or both such fine and imprisonment.

**Section 9.** This ordinance shall be in force and effect from and after its passage.

### Naval Aviation and Latin America

Once more Naval Aviation has scored in furthering the sound relations between the United States and our Latin American neighbors to the South, and in establishing the interests shown by those American Republics in mercantile aviation.

Recently Governor Balbo, of Lower California, and his private secretary, Alfonso Solis, visited the Naval Air Station at San Diego and were given a flight in the Navy's airship B-15 over the beautiful city of San Diego and along the adjacent shore. The government made from his office a statement to the effect that the two men were impressed by the importance of commercial aeronautics, particularly in a country with the difficult terrain of Lower California. Accordingly, he expressed the keenest pleasure in the ceremony shown him by the officers at San Diego, and he stated that, in his opinion, the establishment of commercial aviation in Mexico and in Lower California, at least, would be successful in the near future.

The visit of Governor Balbo to the Naval Air Station at San Diego is in line with previous visits of the same nature to other Central American governments. The frequent visits of naval fliers from the Naval Air Station at Coco Solo, C. Z., to cities such as Balboa, Panama, and to the Canal Zone in the hope of securing of at least one commercial air line in the Republics of Colombia and the projected establishment of others in Panama, Costa Rica, and Nicaragua. And, too, the astonishing performance of the airplane squadrons of the Pacific Fleet, which flew from San Diego to Balboa in the recent joint fleet maneuvers South of Panama and along the West coast of South America, together with the depth of the

airplane divisions of the Atlantic Fleet from Hampton Roads to Panama and Balboa, have brought assistance to the official efforts of all the Latin American countries along the flight route.

The Navy in these recent performances has not only proved the value of long-distance reconnaissance flights in connection with the Fleet, but has given a further demonstration of the practicability of large aircraft for commercial traffic.

### A Letter

#### Editor, AVIATION AND AIRPORT JOURNAL

In the editorial action of the February 25 issue of your publication I have noted an article headed "Economics in Stress Analysis," which is an exaggeration and overstatement of the true situation.

The writer fails to explain the introduction of the theories of Least Work into the structured analysis of airplanes, but I doubt if he can name a airplane designer who did not consider as his first consideration, when he had to design an airplane, this method. The writer is in error when he states that it is an ordinary one. It is like objective to never place less stress a representing computations. The objectives may be sound, but nobody is interested.

When he comes to discuss the so called "Berry method," has no reference of the writer's subject of reduced methods of stress analysis, but the writer's article suggests that he has knowledge of the "Berry method" in respect to what may be gained by means of a glance at a table of Berry functions, etc., he could not have remarked that the "Berry method" is "a" as shown us how to compute stress due to effect were past. Stark's statement is not in the past and loads one to conclude that the writer has not even heard of the "Berry method" which has already occupied him. Every engineer knows that in calculating the stresses on a continuous beam which is subjected to both lateral loads and concentrations and loads the ordinary Three Moment Equations as not often sufficiently precise and even as the unsafe side, and he must turn to the more advanced General Theory of Three Moment Equations for the solution of the more loads loads. Mr. Arthur Berry has arranged this general theorem in a convenient form and has added a table of functions which amount to a series of coefficients to express the elastic properties of the beam which may be introduced into the ordinary everyday Three Moment Equations. A table of Berry functions is a great convenience to a load engineer, and the time required by a competent engineer to use Berry's stock out to the General Theory is negligible, and the results of his work are now more reliable.

I used also disagree with your editorial writer's conclusion that the present state of the industry, when he has no means of a given type are available, is in the best interest of the public. The writer's article is a good example of this conclusion, in a particularly bad form, as it is based upon the importance of commercial aeronautics, particularly in a country with the difficult terrain of Lower California. Accordingly, he expressed the keenest pleasure in the ceremony shown him by the officers at San Diego, and he stated that, in his opinion, the establishment of commercial aviation in Mexico and in Lower California, at least, would be successful in the near future.

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## The New French Aviation Subsidies

Some detailed information is now at hand with regard to the new French aviation subsidies which have come into force on March 1.

**Subsidy for Purchase.**—The subsidy consists of a grant by the State of half the value of machines acquired for March 1, 1931, or of 30,000 francs. This grant is limited to 100,000 francs per year, and the French government has agreed to cover some aviation, applies to the fiscal year 1931-32 and will, as previously stated, be last year.

The subsidies to be granted are of two kinds—

- (a) A subsidy for purchase, and
- (b) A subsidy for public transport.

In the first case, the subsidy will be granted, subject to the approval of the Under-Secretary of State for Air, to French subjects and companies employing French material and French pilots and personnel.

The total effect of the new regulations will be considerably to increase the income of the industry. The purchase subsidy, as also the other subsidies for depreciation, new and general, can be obtained without carrying any commercial load.

#### Method of Allocation of Subsidies

**Subsidy for Purchase.**—The subsidy consists of a grant by the State of half the value of machines acquired for public transport by air, and will not be given to any business connected, directly or indirectly, with the sale of aeronautical material.

The types of machines to be subsidized and the value thereof will be fixed by the Under-Secretary of State for Air. No subsidy will be given to aircraft used by April 1, 1931, or to aircraft imported before Jan. 1, 1931, which are not in the condition to be used, will receive subsidies reduced in proportion to the number of hours already flown.

The companies receiving a subsidy must undertake (1) that he will keep the machines in good condition and repair to the Under-Secretary of State when it is unreasonably for any reason, (2) that the machines will leave French territory within the period of the Under-Secretary of State, (3) that he will sell all the machines without the consent of the Under-Secretary of State.

**Subsidies for Transport.**—The subsidies for transport are granted only to legally constituted French companies, carrying out regular air services of recognized utility. The value of the subsidy will depend on the nature of the subsidies, the type of material required, the regularity of the service and the aeronautical turnover.

(a) For time and distance:  $\text{P} + \text{f} \frac{1}{2} \text{P}$

$$\text{G1} \text{ Depreciation } \frac{1}{2} \left( \frac{\text{P} + \text{f} \frac{1}{2} \text{P}}{200} \right) \text{ for land machines}$$

$$\frac{\text{P} + \text{f} \frac{1}{2} \text{P}}{150} \text{ for seaplanes}$$

$$\text{G2} \text{ Crew } 0.30 \left( \text{Pm} + \frac{\text{E}}{2} \right) \text{ for land machines}$$

$$\frac{\text{E}}{150} \text{ for seaplanes}$$

$$0.40 \left( \text{Pm} + \frac{\text{E}}{2} \right) \text{ for seaplanes}$$

$$\text{G3} \text{ Transport } \frac{\text{E}}{2} \text{ for land machines}$$

$$\frac{1.5 \text{ E}}{1000} \text{ for seaplanes}$$

$$\text{G4} \text{ Gasoline } \frac{1000}{0.250 \text{ mHP}} \times \frac{\text{P}'}{0.65} \text{ for stationary engine}$$

$$\frac{1000}{0.250 \text{ mHP}} \times \frac{\text{P}'}{0.65} \text{ for rotary engine}$$

$$\frac{1000}{0.65} \text{ for land machines}$$

$$\frac{1000}{0.65} \text{ for seaplanes}$$

$$\frac{1000}{0.65} \text{ for stationary engine}$$

$$\frac{1000}{0.65} \text{ for rotary engine}$$

Where  $\text{P}$  is the price of the aircraft as fixed by the Under-Secretary of State for Air.

Where  $\text{P}'$  is the price of the engine as fixed by the Under-Secretary of State for Air.

Where  $\text{E}$  is the price of the gasoline, to be fixed quarterly. Where  $\text{E}$  is the average length of a stage on the route.

Where  $\text{f}$  is the horsepower.

Where  $\text{mHP}$  is the horsepower.

Where  $\text{m}$  is the coefficient depending on the average length of a flight and on whether the route is entirely in France, crosses over French Africa, or is international.

Where  $T$  is the number of tons of commercial load carried.

Where  $\text{P}'$  is the speed in kilometers per hour of 2,000 meters.

(b) **For Transport.**—This subsidy will be granted at the rate of 975 francs per kilometer per kilogram of goods, carried at fixed rates by the Under-Secretary of State for Air.

#### Special Conditions Governing Grant of Subsidies

The company must possess a nucleus of machines equal to twice the cost obtained by dividing by 200 (in the case of land machines) and 350 (in the case of seaplanes) the number of flying hours. Represented by the total annual distance flown by the company's machines at a speed of 130 kilometers per hour.

The company must employ as a minimum the following personnel—

(a) One pilot for every three machines.

(b) One mechanic for every 200 kg. utilized.

Where  $\text{f}$  is the number of hours which may be charged by companies and fixed according to the following total load flown by the Under-Secretary of State for Air:

Passenger, 80 francs per passenger-kilometer; Goods, 915 francs per kilogram-kilometer.

**Refugee of War in French Estates.**—In effect, the new subsidies will produce the following reductions in passenger fares (per flying stage, not including handling charges)—

Stage fare Paris-Strasbourg, reduced from 300 to 250 francs.

Stage fare Paris-Prague, reduced from 1,200 to 800 francs.

### Belgian Colonial Aviation

The progress of aviation in the Belgian Congo is remarkable, judging from recent reports. Thanks to the financial support of King Albert, who is himself an enthusiastic aviator, a flying service, half Belgian for the Congo Colony in Oct., 1930, and the first flight actually took place there in March, 1930. Since that date a regular flying service has been in existence between Kinshasa and Oyo, and, in the vicinity of Oyo, between the stations of Tshinga. An extension from Kinshasa to Luluia will be inaugurated during the month on the arrival of the next steamer, and the service from Luluia to Stanleyville, a distance of about 300 miles, will be working before the end of this month. Flying boats are being employed, the course of the Congo being followed as far as it is possible. Up to now no regular air service exists between Oyo and the upper river route on the Congo. For the mail service, the upward voyage taking twenty days, the return occupying fifteen days. When flying regularly it is estimated that hydroplanes will not exceed three days on the trip. Up to the present a reply has not been received from the Congo, but it is expected a delay of three to four months.

While the service has been suspended, the Belgians are able to use the services of the mail steamer which has served the original routes. Great importance is attached to the development of aviation in the Congo and the Societe Financiere, which controls the seasonal routes in the Kasai at Dyck-Pandy, is preparing to establish an air service between Stanleyville and the town of Luluia.

When the new hydroplane is ready to go to the Congo, a hydroplane service of this river Congo has also been desired upon, and it is hoped to complete the work by hydroplanes in about three years time, as against six to fifteen years in the ordinary way.

The energy and intelligence exhibited by the Belgians in the development of their colonial aviation are altogether remarkable.



## General Message on Commercial Aerostatics

Commenting on the National Southern Air Tournament of Bellair, Florida, Maj. Gen. Charles T. Menster, Chief of Air Service, said:

"The Air Service sees in commercial aerostatics a solution at least one great problem confronting the world today. The nations referred to in the war was a progressive economic measure and a more or less undefined fact that security in the future was even more necessary than in the past."

"The development of the airship has now demonstrated to the satisfaction of military leaders that the air controls both land and water; that the first (and possibly the last) battle of the next war will be fought in the air. If this is true, and I believe that it is, then the airship would seem to be the most potent weapon ever produced by man. At the present time the forces of the airship begin and ended with its application in commercial aerostatics, which appear to be insuperable rather than an obstacle to progress."

"But this is not all. The use of aircraft during the conflict placed aerial navigation before the eyes of all people and opened up wonderful possibilities. The airplane offers to commerce in a constructive capacity than it does to the Army or the Navy in a destructive. We present the fastest means of transport, the most accurate planes, fighters. It carries people and freight great distances, over land and through, increasing the range of influence of the world's leaders, and thus goes far toward eliminating that steamer and more dangerous friendly, which has often been the heaviest contribution to international difficulty—the frontier of ignorance and猜度。

"The time is not far off. It is already in process of realization. Already is the American flying Corps sent to cover 20 hours and one of our military pilots has conquered the same distance in 22 hours. The air has brought us within range of our territory in Alaska on the north and the Panama Canal on the south. It has placed Europe much nearer and the Air Service Japan, before long, will see the Pacific flown in a single day. The development of the airship is the state of the art that is insuperable, but in a week of "stop" from California to Hawaii, to Guam and thence to the Philippines."

"When this is accomplished, the unity of mind between America and Asia should be increased, for with closer contact and a mutual attraction to continental ends, there is certain to be a better understanding and lessened opposition. The nearer West and East are brought nearer, the sooner will insuperable differences be seen as possible of adjustment."

"So much for commercial flying as directly applied to business. Its relation to the national defense follows as a matter of course. No nation can hope to realize sufficient military strength to control the air unless it can also control commercial and national development. It is to this that there can be no security without aviation, does it not follow that there can be no aviation without commercial aerostatics which provide a reserve in production facilities and in trained personnel."

"Of all the wonderful developments credits to the world war, the aircrafts do not hold an argument when placed side by side with the productive activities of the post-war period. In this position in future economic phases nations even more certain to civilization as a dominant factor in the next war. The Air Service consequently feels that the advancement of commercial flying—its encouragement, control and guidance—is a national duty. Aerial tournaments should greatly stimulate public interest and knowledge and consequently hasten adoption of a sound aerostatic policy."

## Notice to Aviators, 1921, No. 3

Notice to Aviators, 1921, No. 3, issued by the Hydroaeronautical Office of the U. S. Navy, lists information on aviation facilities available at the Aerostatic Field at Bellair, St. J., on Biscayne Bay; at Hampton Roads and Chesapeake Bay; at Corpus Christi, Texas; at Fort Bragg, North Carolina; and Redding and McWayne, in California; and at Mile Bay, H. T. and Molokai Islands, H. T.

Information is also given on the daily forecasts of flying weather conditions which the Weather Bureau issues.

## New Contemporaries

The steady growth of aerostatic efforts all over the world is well illustrated by the number of new aviation magazines which have appeared in the last year. Some of these existed before the Great War, but they had to suspend publication while the conflict lasted. Now they are resuming again and many new magazines have joined their ranks.

AVIATION AND AVIATION JOURNAL recently received enough copies to illustrate the appearance of three such publications in a series of countries. These are: "Aviation" is sold in Austria; "Aviation" is the magazine of the respective aviation in Ireland; the official organ of the Czechoslovak Aero Club, and is published monthly at Vyškovice, Prague, Czechoslovakia. Another is *ASIA*, the official organ of the Imperial Asian Society of Japan, with publication offices at 1, Tamagawa, Kita-ku, Tokyo, Japan. This magazine, which is evidently printed in Japanese, announces itself with the following statement:

"We have the honor to inform you that the civil aviation in Japan is going to develop very rapidly, so that we have had to import foreign machines, motors, and accessories, which should be exchanged. The present situation being like this, we have decided to establish the 'Asia Aviation' and 'Asia Aeroplane' which is the organ of the Imperial Aero Society of Japan representing the Empress's civil aviation in the Imperial Aeroplane Federation, and it is the only best and most popular Magazine in the Empire, and we have now about thirty thousand readers among which are aircraft manufacturers, army and naval officers, business men, scientists, scholars, etc."

The third newcomer is in the field of aerostatic news in *AVIATION*, of Tucuman, Argentina, Buenos Aires, Argentine Republic. This magazine, which is published monthly, is the official organ of the Centro Aviacion Civil, a civilian organization which has for its purpose the development of Argentine aviation through training and propaganda work.

To all the new organizations, AVIATION and AVIATION Journal extends its best wishes for success and prosperity.

## A. C. A. Open Club House at Flying Field

The Aero Club of America has accepted the offer of the Cecilia Co. to use up to 100,000 square feet of Hangar space for the use of the Aero Club of America. The Club House will be ready by the 15th of May and will be specifically arranged for the convenience of the members of the Club and their guests, most of whom have arranged for keeping at their stables at the field, and accommodations will be provided for visiting aviators from the different states of the country. Arrangements are also being made so members of the Club and their guests may take flights in the aircraft in cooperation with the government, have the use of government machines.

## Change of Address

H. G. Smith Tool and Mfg. Co., manufacturers of laboratory tools and apparatus, factory of 315-317 Market St., Newark, N. J., have moved into a large and convenient factory located at 245-247 N. J. E. Ave., Newark, where they will engage on a large scale in the manufacture of their recently developed Standard Radon Lathe and Planer Tool. Each of these includes the excellent Smith "wetting-off" bed and threading tool. The company is also fully equipped for the manufacture of special tools, metals, dies, rigs, gauges and fixtures.

## Balloon Club Proposed Near New York

A group of lighter-than-air men is desirous of establishing in the vicinity of New York a sporting club provided with a balloon field. The members of the club would be limited by no means. No far-off tentative discussions have been had with regard to this subject, but it is believed that a recently perfected process of generating good gas at very low cost would make such a club a successful proposition. Lighter-than-air men interested in the matter should communicate with Ralph H. Upson, 22 East 17th Street, New York.

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